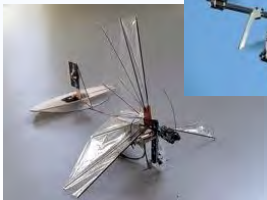


# Trends in UAV Propulsion

Liban Emanuel

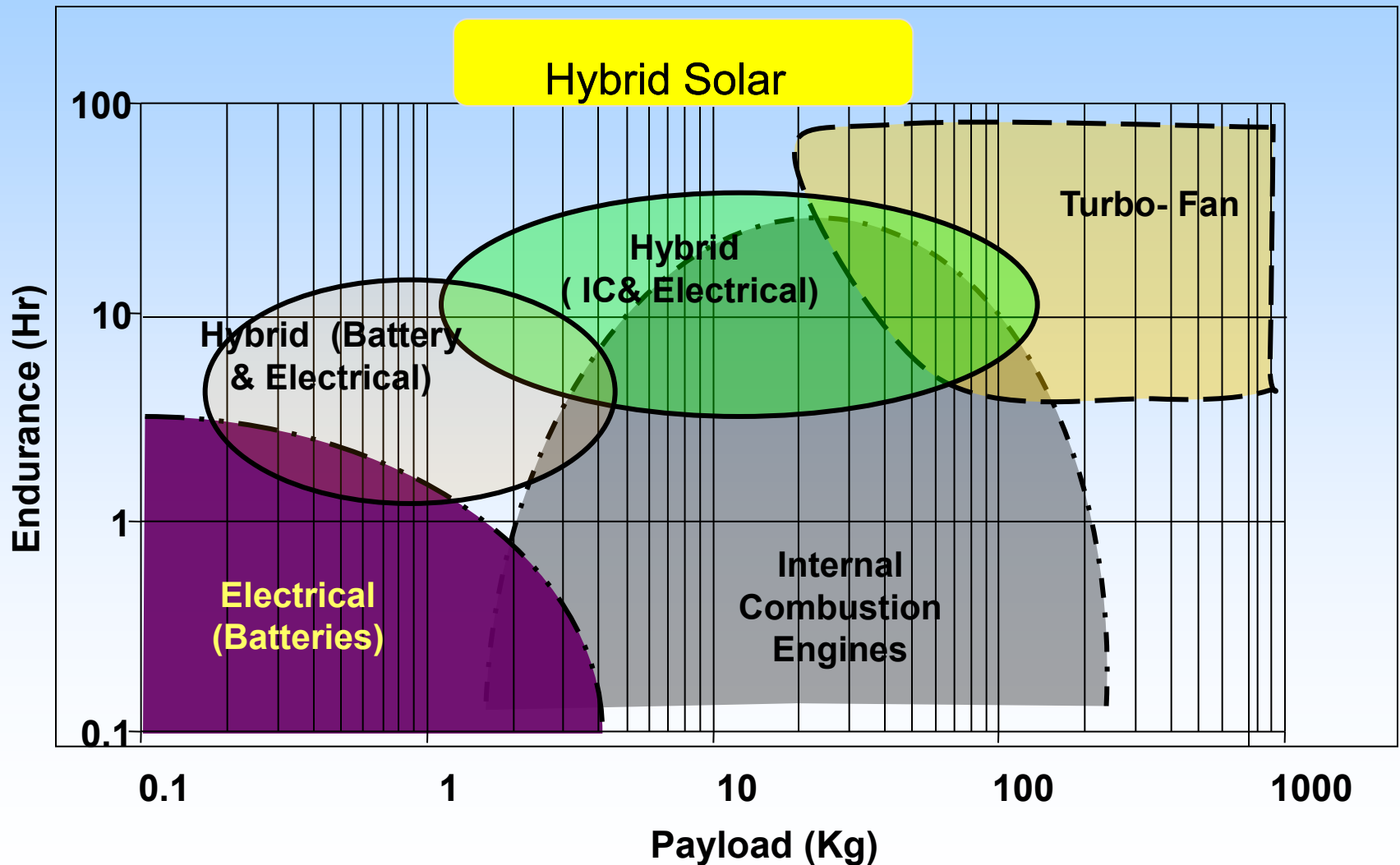
# Trends in UAV Propulsion



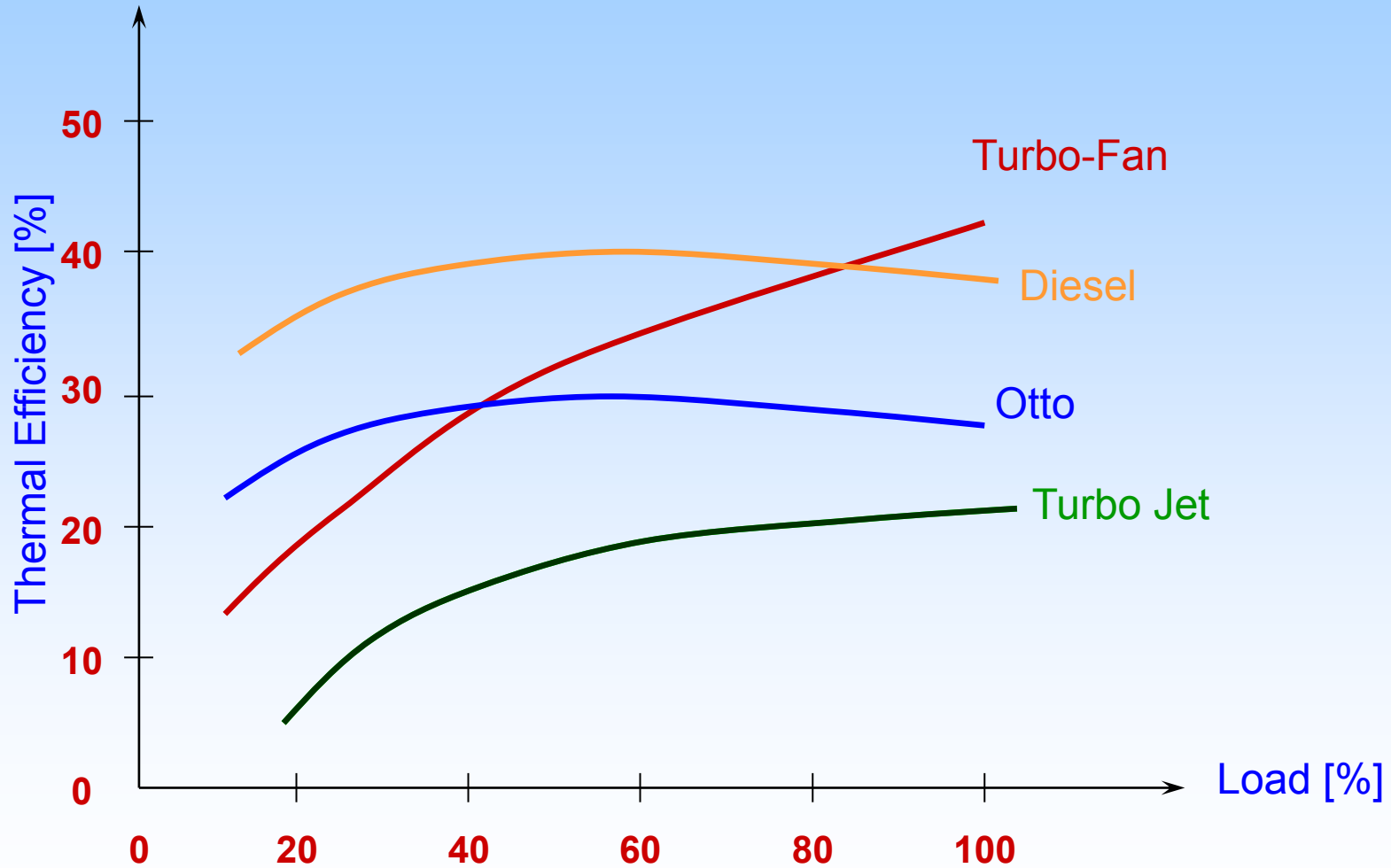
# UAV Propulsion Requirements

- ❑ **Reliability**
- ❑ **High power to weight ratio**
- ❑ **High efficiency in the flight envelope**
- ❑ **Low emission of IR ,Radar and Noise signatures**
- ❑ **High Electrical Output for the UAV function and payloads**
- ❑ **Low LCC**
- ❑ **Compliance to Civil Aviation Regulation**

# Domain of usage of different propulsion technologies



# Relative efficiencies of different engines



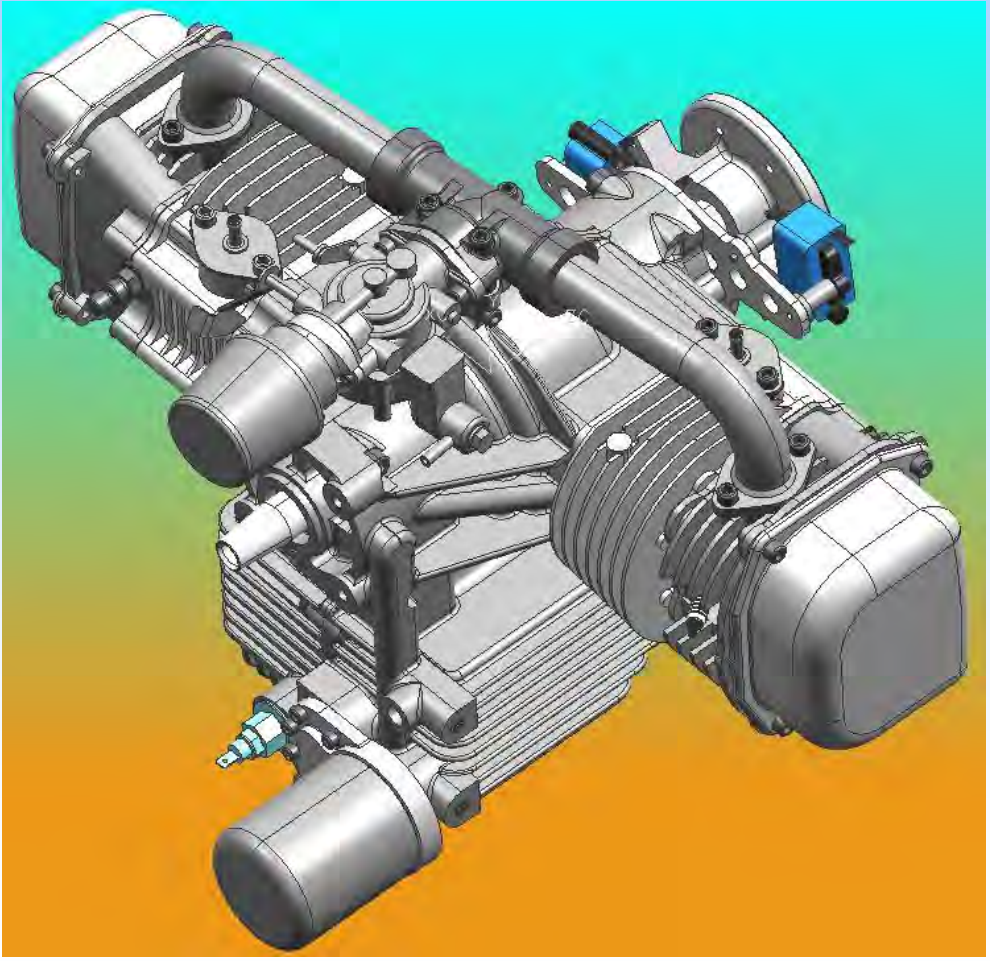
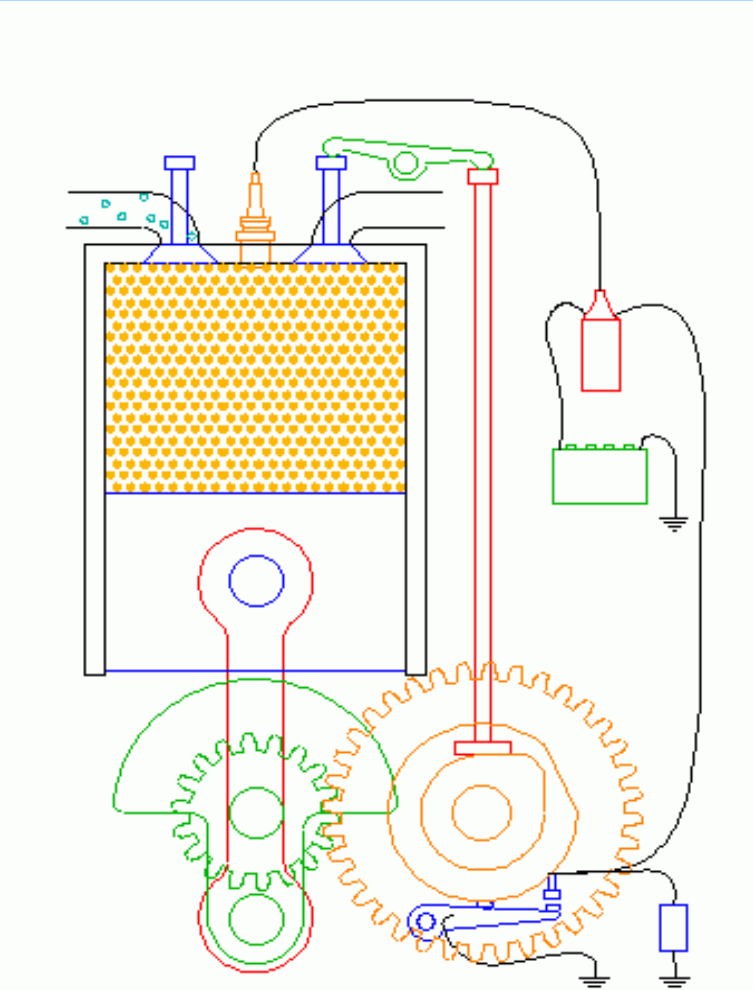
# IC Engines Benchmarking

Feature	LCC	Noise	MTBF	Life	Fuel consumption	Power to weight Ratio	Weight	Instalation	Simple
Two Stroke (2)	++	++	++	++	+	++++	++++ (5)	++++ (5)	++++ (5)
Four Stroke	+++	++++	+++	+++	+++	++	++	+++	++
Wankel (1)	+	+	+	+	++	++++	+++	+++	++
Diesel (3)	+++	+++	++++	++++	++++	+	+	++	+

1. Expensive and unique parts
2. Simple and cheap
3. Expensive ,Exellent SFC &LCC,Heavy
4. Advantage for low HP and Air- Cooling.

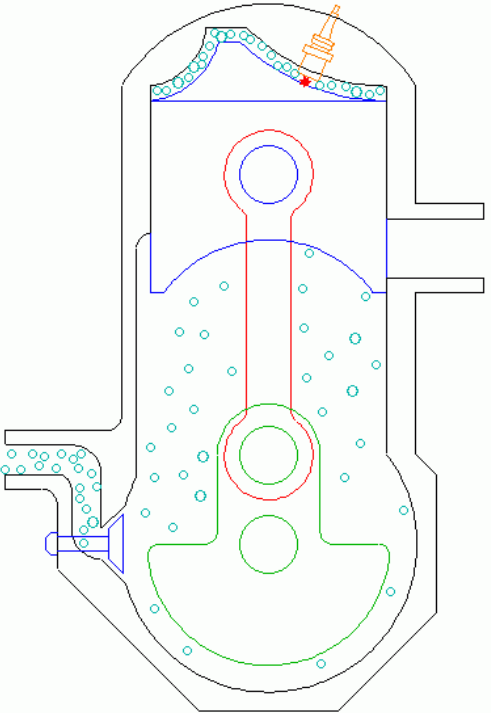
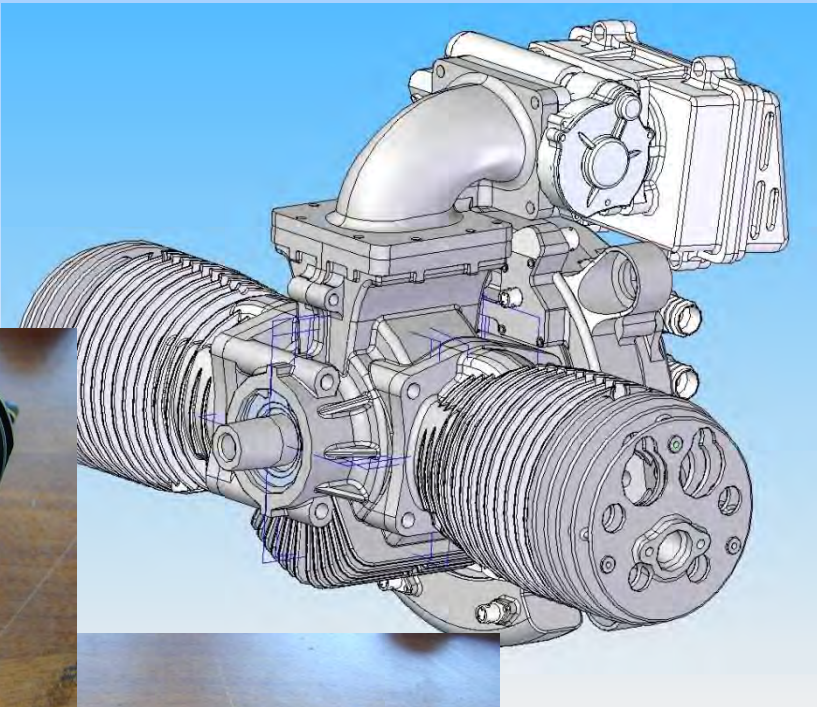
Legend: +++++ The best

# Four Stroke Engine



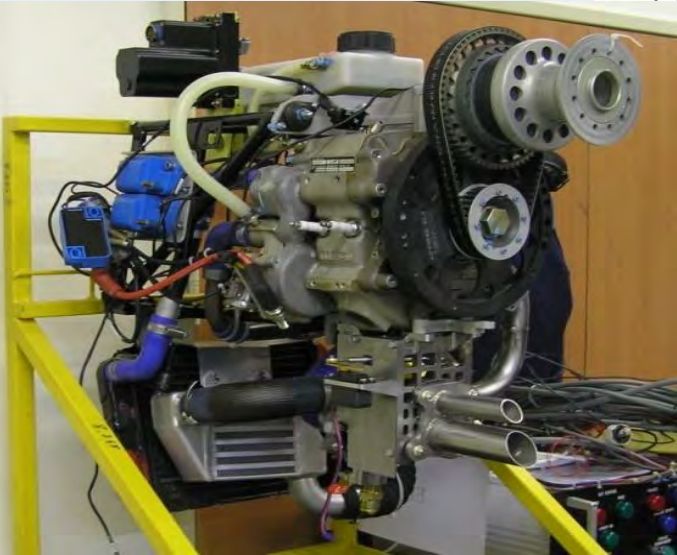
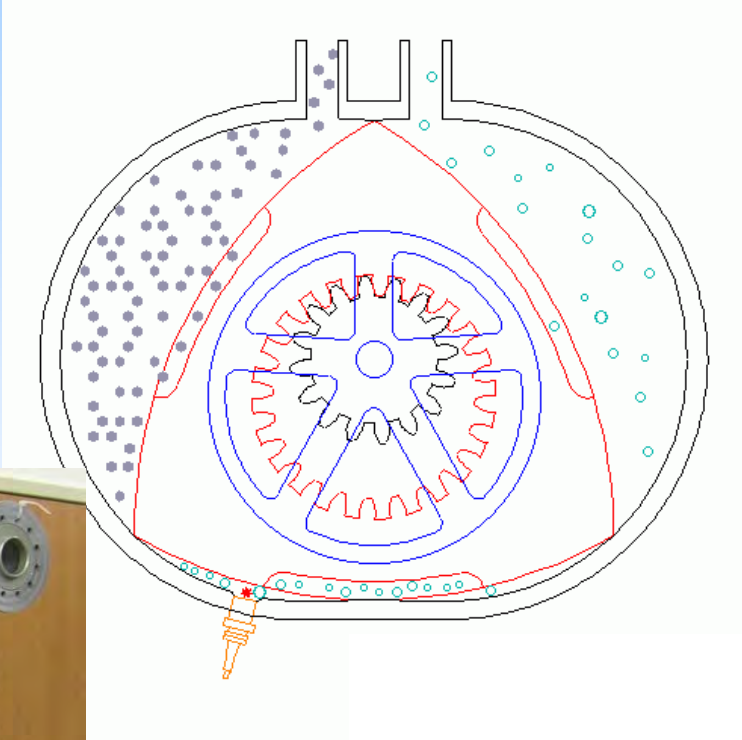
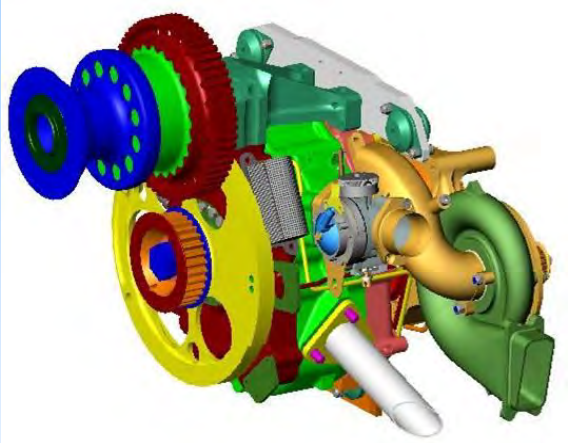


# Two (2) stroke engine





# WANKEL Engine



# IC Turbo - charging



*BMW- Tri Turbo Diesel engine*

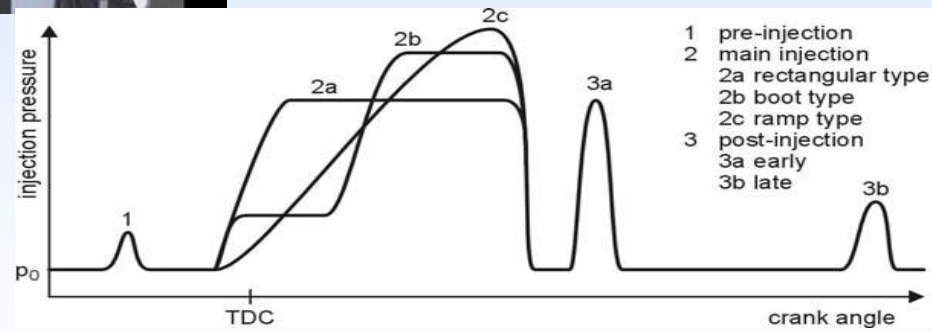


*Volkswagen TSI engine*

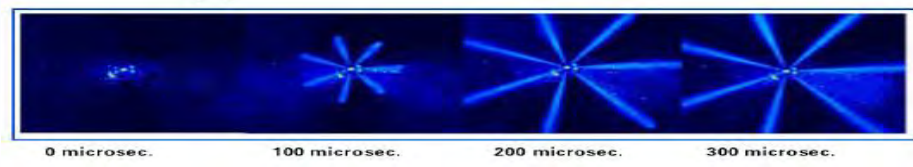


*VanDyne- Turbo compound*

# Direct Fuel Injection



**Very-High pressure  
 fuel system**

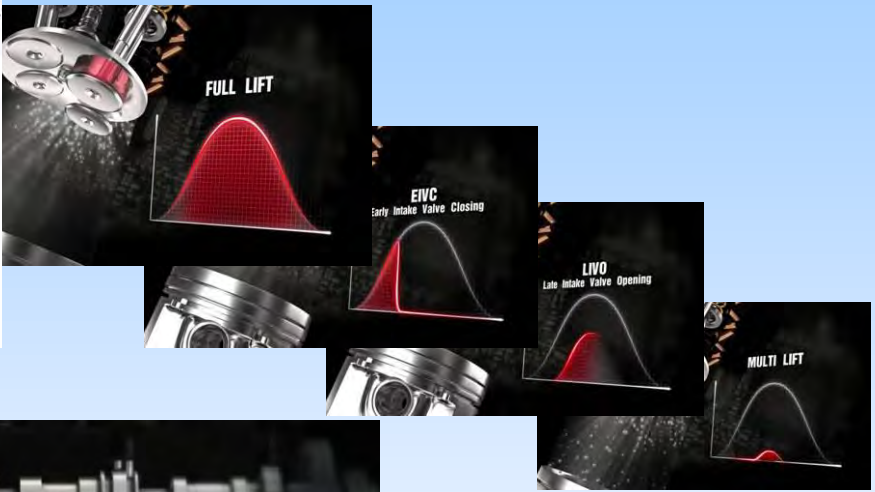




# Valves control (Semi -Atkinson Cycle)



*FIAT Multiair*



- ✦ Active valve opening control (Atkinson cycle)
- ✦ Adaptation of thermo – cycle to loads and power needs.
- ✦ 5-15%Efficiency gain

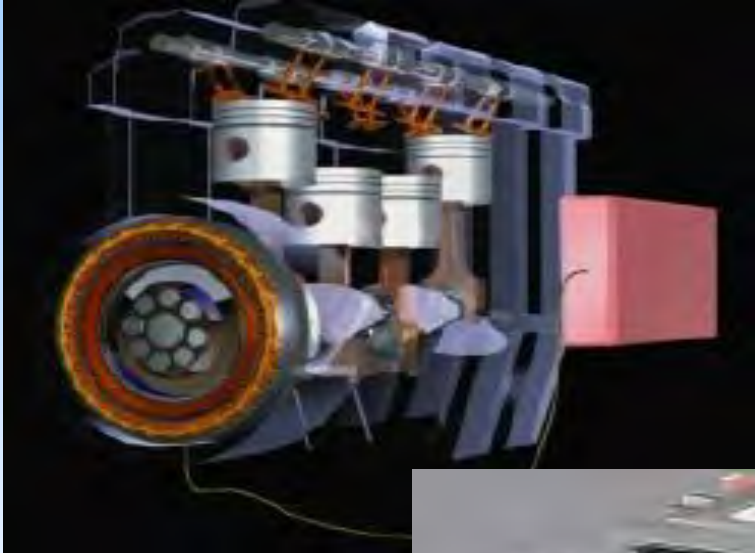


**1,4 I TSI Zylinderabschaltung (ZAS)**

ZAS inaktiv	ZAS aktiv
Nockenhubumschaltung: Vollast- / Teillastnocken	Zylinderabschaltung (Zyl. 2+3): „Null-Nocken“ = Grundkreis
	
Ventilhub Zyl. 1-4	Ventile Zyl. 2+3 geschlossen



# Starter - Alternator



- ✦ Efficient
- ✦ High Power and Moment
- ✦ “Built In” structure
- ✦ Reliable
- ✦ Enables Hybrid Propulsion

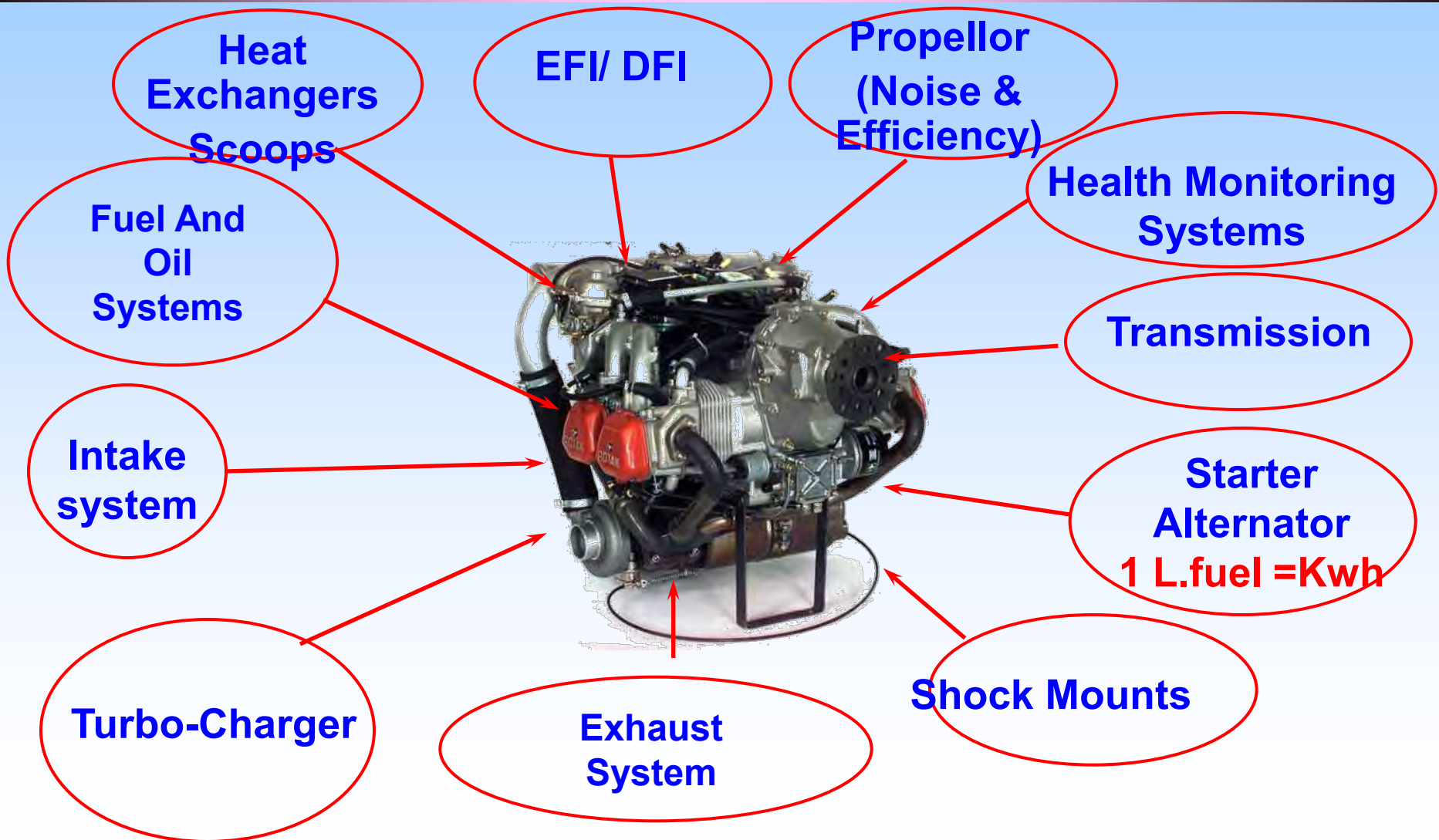


# Electric Wheel Drive



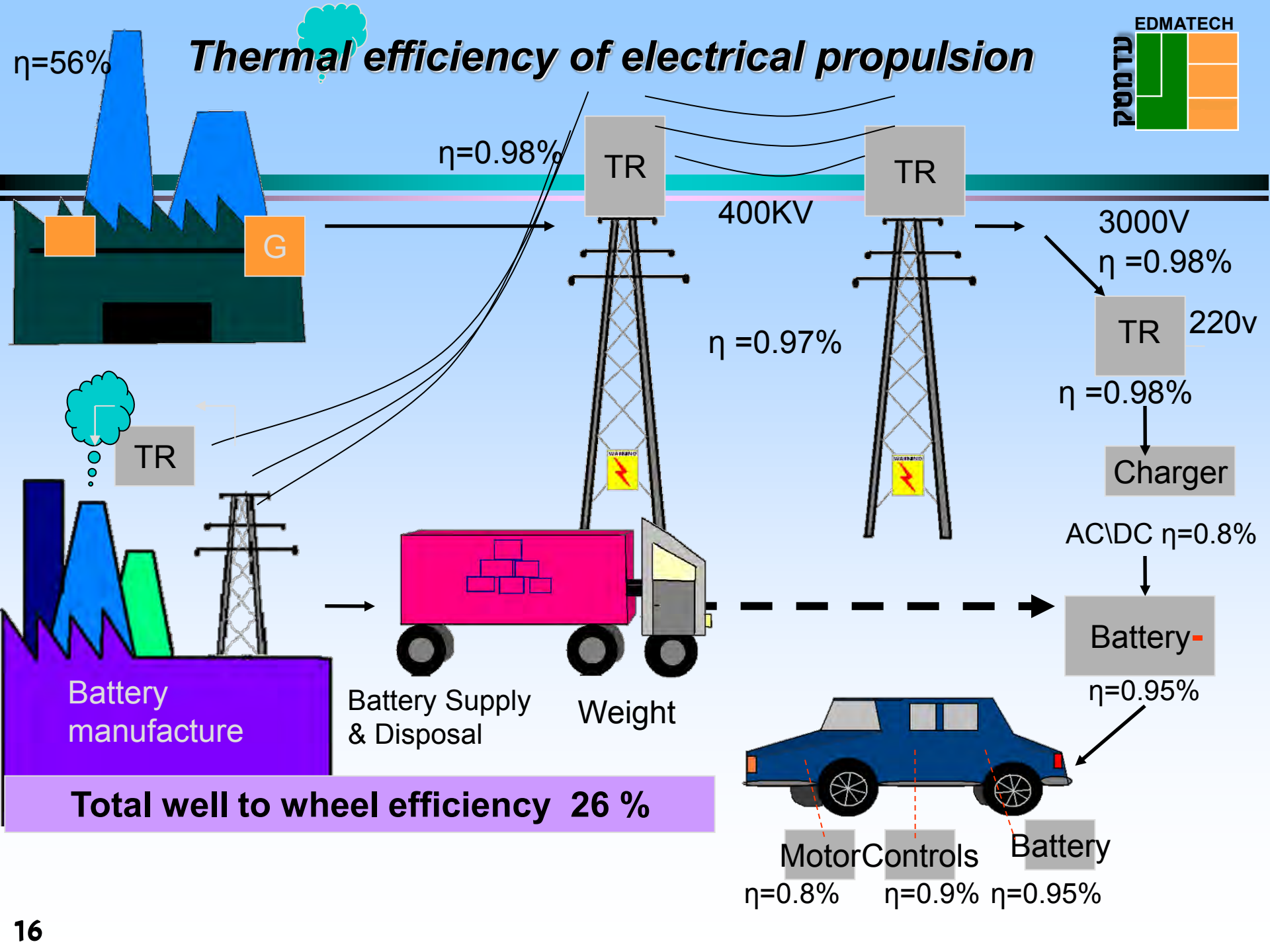


# IC – UAV Systems



# Thermal efficiency of electrical propulsion

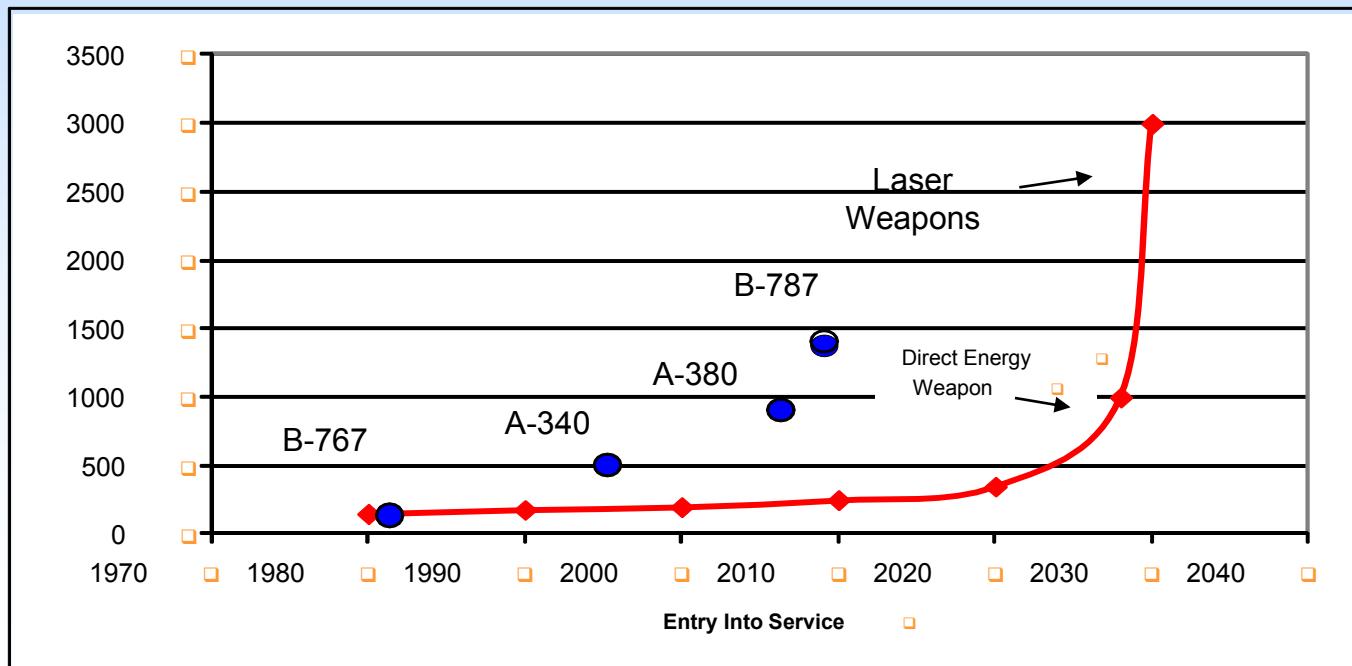
$\eta=56\%$



**Total well to wheel efficiency 26 %**

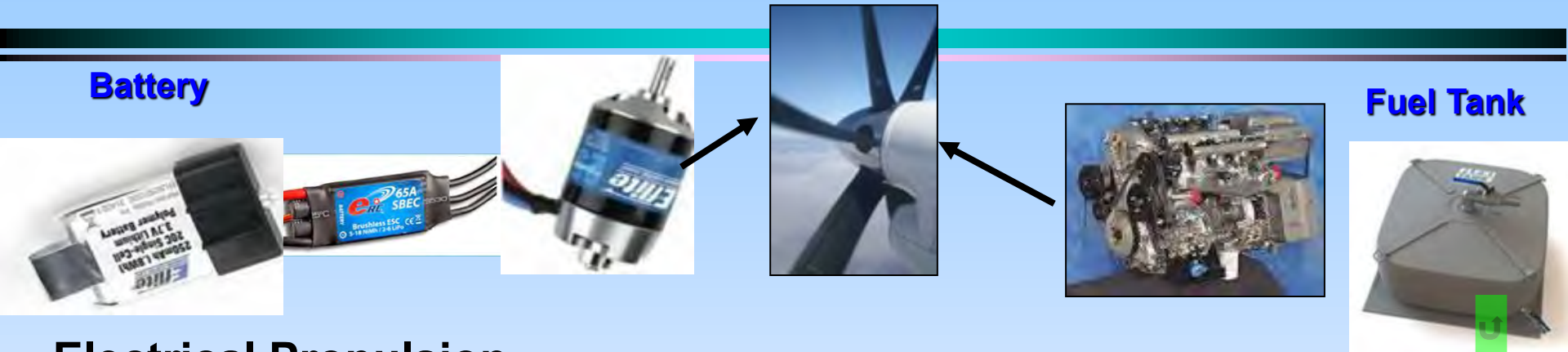
# Trends – Electrical propulsion

- ✘ Efficient high -torque electrical motors with power to weight ratio of 8+
- ✘ Increased demand for a/c electrical consumption for payloads.
- ✘ "Intelligent" control of power sources and usage
- ✘ High voltage systems
- ✘ Batteries / Fuel cells with power to weigh ratio 300 --- 600 Watts/Kg



(KW) □

# Energy Density - Elec. Drive Vs. IC Engine



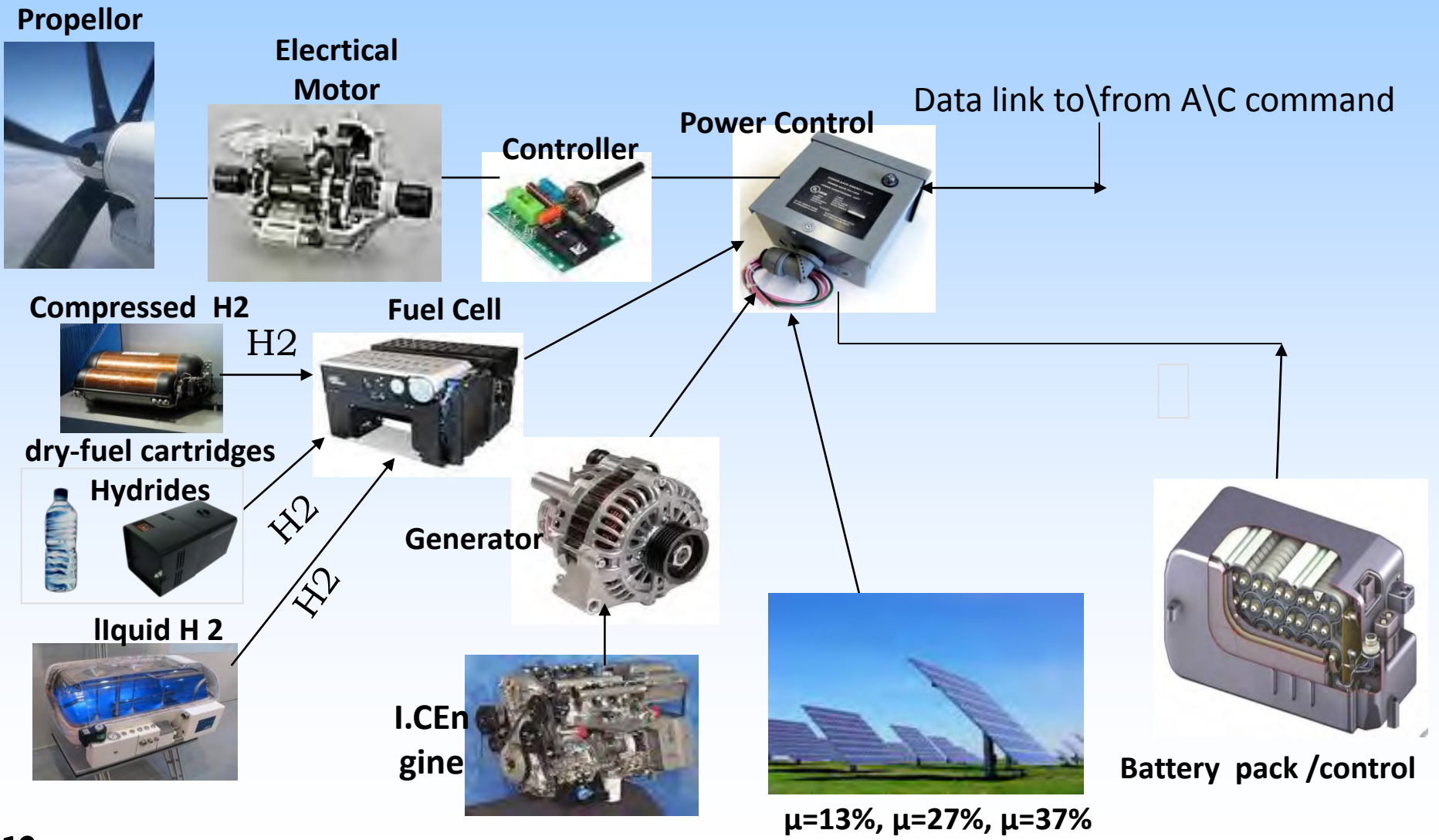
## Electrical Propulsion

- ✘ Battery Energy density – 0.2 kwh/kg
- ✘ Controller Eff. = 0.9
- ✘ Motor eff. = 0.9
- ✘ Energy Output:  
 $0.2 \times 0.9 \times 0.9 = 0.16 \text{ kwh/kg}$

- ✘ Fuel Energy Density - 12 kwh/kg
- ✘ Efficiency: Otto= 0.3
- ✘ Efficiency: Diesel= 0.4
- ✘ Energy output:
  - ✘ Otto -  $12 \times 0.3 = \underline{3.6 \text{ kwh/kg}}$
  - ✘ Diesel -  $12 \times 0.4 = \underline{4.8 \text{ kwh/kg}}$

**The ratio between Energy density of IC/ Fuel and Elec. Battery/Motor is 20–30 to 1 !!**

# UAV IC / Electrical hybrid

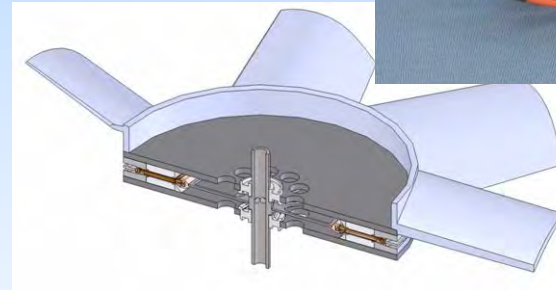
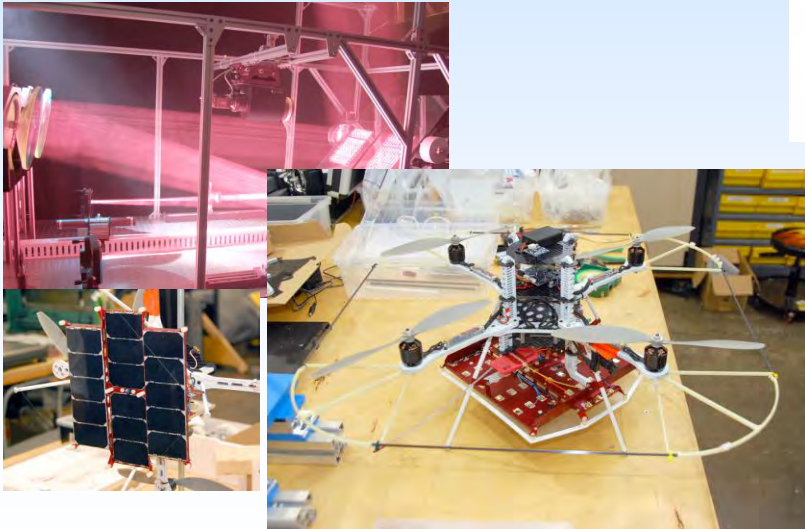




# Electrical Propulsion

## Laser-- Radiated Energy ,HF

- ✦ Very low efficiency (3.5%)
- ✦ “At starting point”



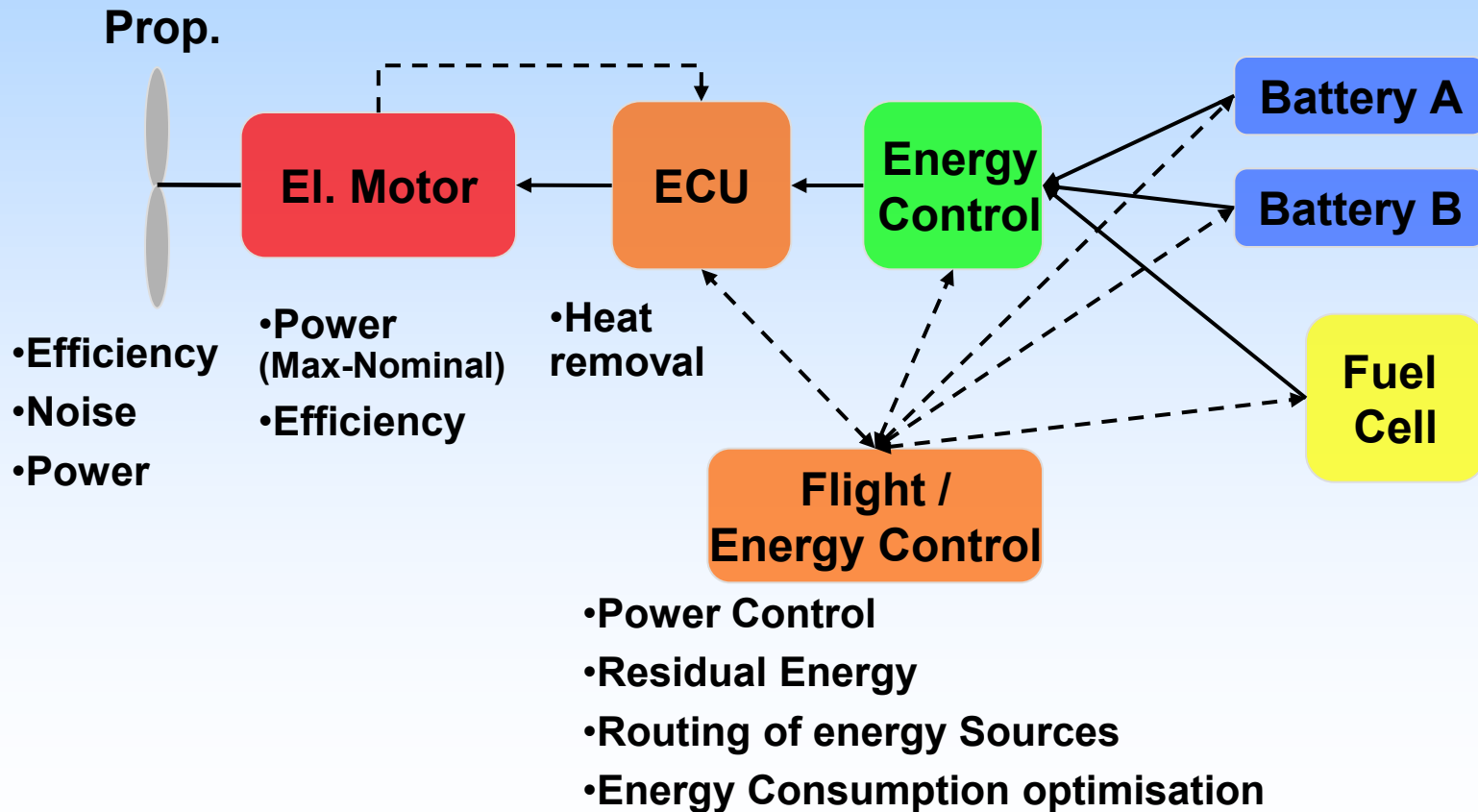
## Elec. Motors

- ✦ Weight to power ratio 1:8 Hp/Kg
- ✦ Efficiency (0.95)
- ✦ Axial Flux/ Halbach



# System Consideration

- ✧ Motor /Propellor optimisation
- ✧ Max. Efficiency @ Design point
- ✧ Optimal Useage and Control of Energy Sources



# Fuel Cells for UAV 's

## □ Trends

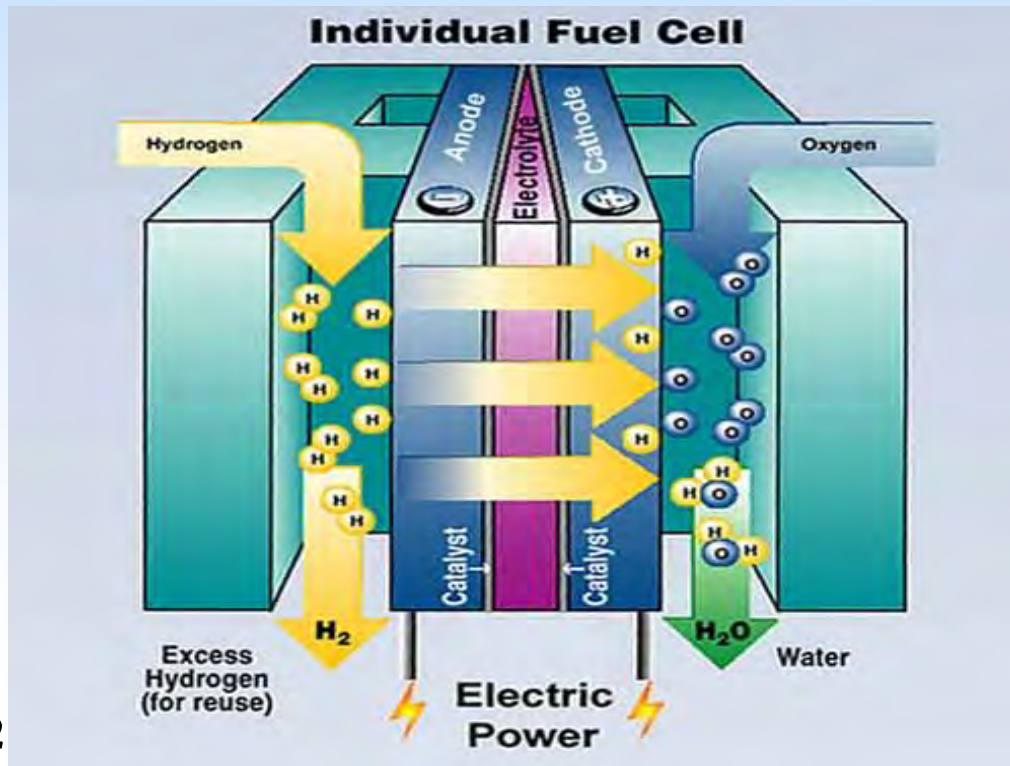
- Hydrogen storage improvements
- Onboard Hydrogen generation
- Use of Methanol
- “Closed loop” water system

## □ Advantages

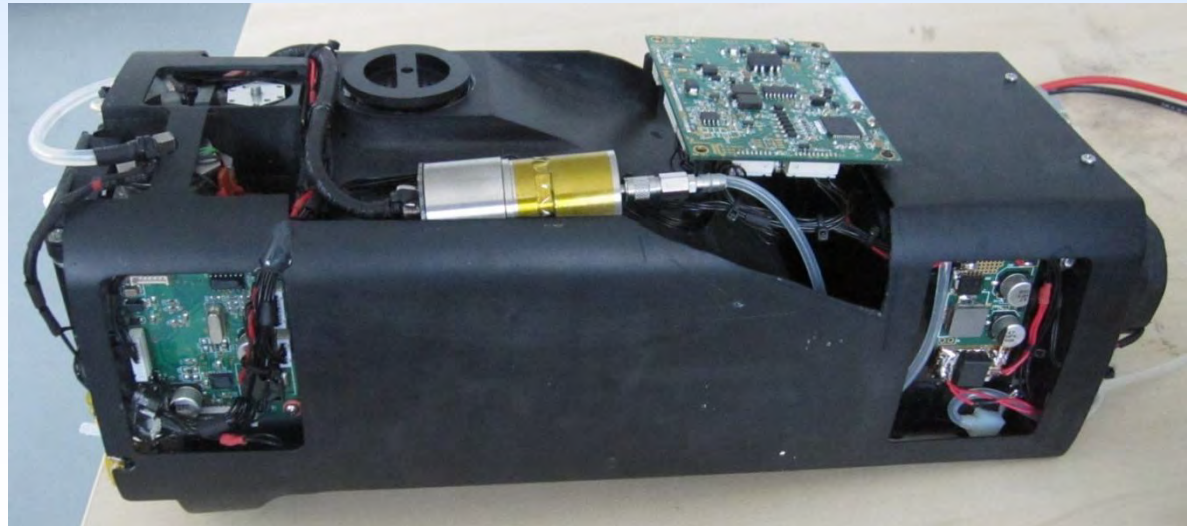
- Higher Energy Density than battery
- “Fuel Tank Concept “

## □ Shortcomings

- Complexity
- Price.
- Technology not yet mature



# UAV - Airborne Fuel Cell

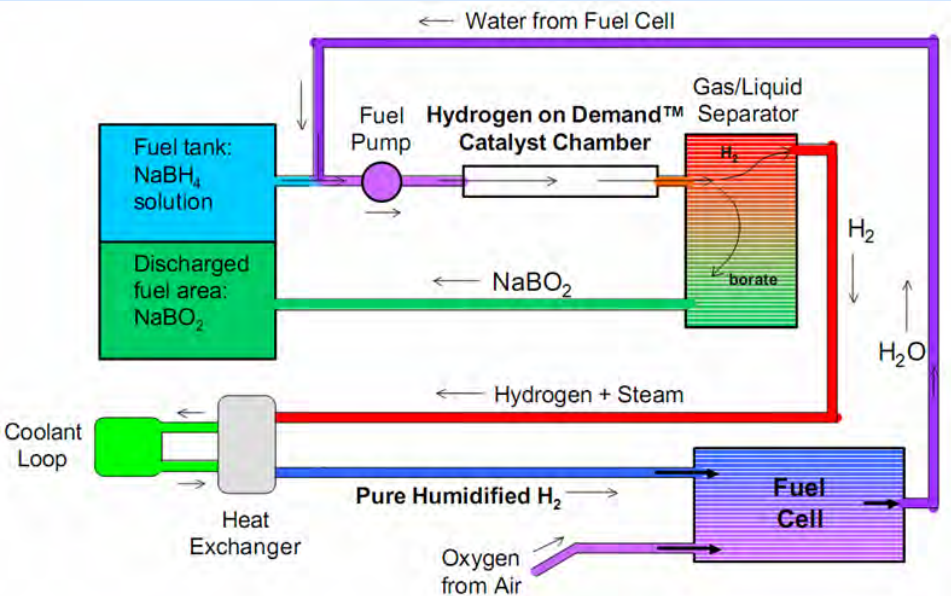


# Fuel Cells Application



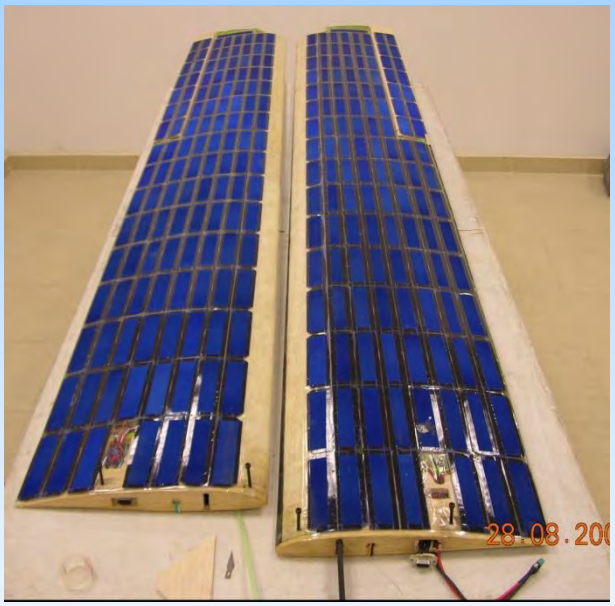
## Technologies

- ✦ PEM
- ✦ Hydrides/PEM
- ✦ Methanol DMFC
- ✦ Propane /SOFC





# Photovoltaic hybrid propulsion



# Hybrid Propulsion





# PV Propulsion

- ✧ **High Altitude Application**
- ✧ **Unlimited Time on station**
- ✧ **Technology of PV cells and batteries are not yet mature for this application**

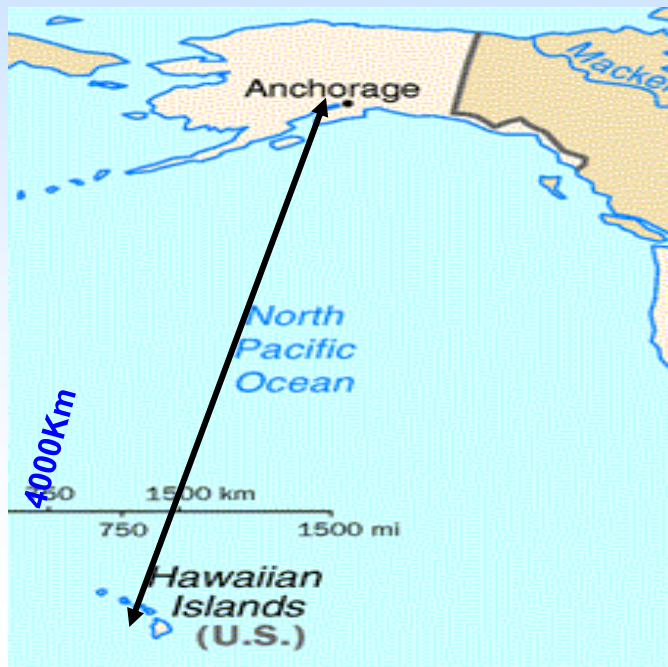


# Bio Propulsion

## Golden Plover UAV

Consumption - 70 gr. of Fat = Fuel

Flight of 4000 km non- stop 88 hrs / 250,000 Wing Flaps



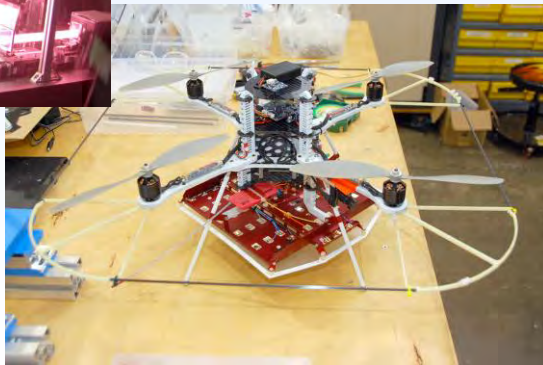
Take-off weight – 200 gr

Landing weight – 130 gr.

# Future Trends

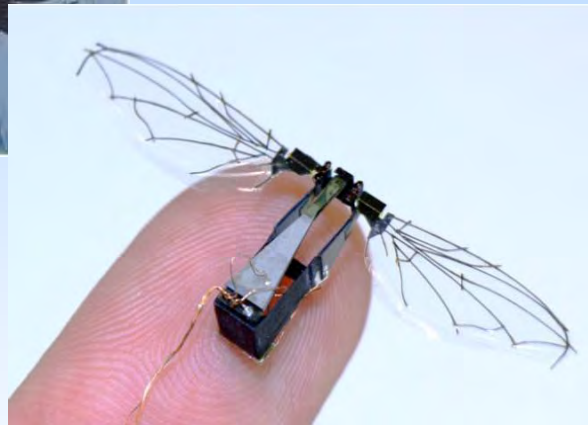


**Albatros**



# “Bio” - Propulsion

We try to understand and learn?



Low Reynolds Aerodynamics

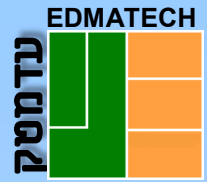
Airfoils

Materials

Smart structures

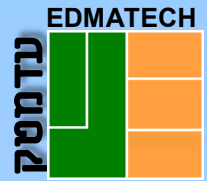


# Summary of trends IC Engines



- ✘ **I.C Engines will continue to dominate UAV Propulsion due to the high energy density/content of the HydrocarbonFuel.**
- ✘ **UAV Engines benefit from the developments in Automotive Industries.**
- ✘ **More and more UAV'S will require Heavy Fuel capability.**

# Summary of trends Electrical Propulsion



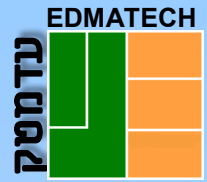
- ✦ “Break Through” in Energy capacity and power to weight ratio of batteries is needed .
- ✦ The portable I.T. devices as well as automotive applications are taking the lead.
- ✦ Usage for small UAV’S with limited flight duration.
- ✦ Important achievements in Electrical Motors and Generators.



# Summary of trends Fuel Cells

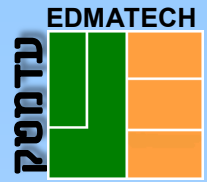
- ✘ Promising technology with higher energy density than batteries  
(600 wh/kg. and up)
- ✘ Bottled high pressure hydrogen with PM are practical solution for small long-range UAV'S.
- ✘ Cryogene-hydrogen systems can increase range and payload.
- ✘ Other Fuel- Cell systems are complex and require considerable effort to mature and to reach reliability.
- ✘ In next year or two we shall see Automotive Fuel cells on the market and maybe these technologies can be used in UAV propulsion.

# Summary of trends -Photovoltaic Propulsion



- ✦ **Need further advances in:**
  - ✦ **Lighter structures.**
  - ✦ **Lighter and higher efficiency of P.V cells.**
  - ✦ **Batteries or Fuel- Cells back-up need much higher power density.**
  - ✦ **Flexible wing flight controls.**
  - ✦ **Efficient Low-speed Propellers.**

# Summary of trends Hybrid Systems (I.C and Electrical)



- ✘ **Important load fluctuations.**
- ✘ **Big difference between the requirement for duration of High- Power (T.O) and the cruise power.**
- ✘ **The A/C design and structure dictates distributed electrical propulsion.**
- ✘ **Automotive Industry is leading.**

# The dream





# Summary

**We still have a lot to learn ...**

**Thank You**